# Comparison of Adaptive and Conventional Backstepping Super-Twisting Sliding Mode Controls for Trajectory Tracking of an Autonomous Underwater Vehicle

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**Abstract:** Maritime missions involving autonomous underwater vehicles (AUVs) often encounter uncertainties and external disturbances, posing significant challenges for trajectory tracking. These challenges complicate the design of controllers capable of achieving high-precision trajectory tracking. This paper proposes a comparison between the backstepping super-twisting sliding mode control (BSTSMC) and the novel adaptive backstepping super-twisting sliding mode control (ABSTSMC) approach for trajectory tracking of AUVs in 6 degrees of freedom (DOF) under thruster fault conditions. The ABSTSMC method retains the benefits of robust control while incorporating adaptive techniques to minimize the impact of unknown external disturbances and thruster faults by adjusting the gain online. The performance of both BSTSMC and ABSTSMC is evaluated and compared through simulation results using ROS2, with a focus on their ability to handle external disturbances and thruster faults.

# Keywords— Autonomous Underwater Vehicle (AUV), Adaptive Backstepping Control (ABC), Super-Twisting Sliding Mode Control (STSMC), Thruster Fault

#### I. INTRODUCTION

Autonomous underwater vehicles (AUVs) are increasingly employed in various underwater missions such as oceanographic surveys, environmental monitoring, military operations, submarine oil pipeline detection, and underwater rescue. These missions often require precise trajectory tracking capabilities in challenging underwater environments characterized by uncertainties and disturbances.

In recent years, various control techniques have been developed to enhance the trajectory tracking performance of AUVs. The efficiency of these controllers is heavily based on various factors, particularly the underlying mathematical model. A prominent example is an influential mathematical model of AUV proposed by Fossen, as outlines in [1-2]. This model also integrates environmental forces, such as ocean currents, into motion significantly impacting AUV equations. control. Designing a controller for high tracking accuracy is challenging because of the uncertain parameters in the AUV model, nonlinearity, and unknown external disturbance. In recent years, various controllers have been developed to solve many trajectory tracking issues for AUVs.



Fig. 1. The Xplorer-mini AUV, equipped with eight thrusters, developed at Autonomous Marine Research Lab (AMARR Lab) at Faculty of Engineering, Kasetsart University

Conventional control methods, such as Proportional-Derivative (PD) and Proportional-Integral-Derivative (PID) controllers, are commonly applied because of their straightforward implementation. However, PD controllers require accurate knowledge of buoyancy and gravitational force [1], while PID controllers face performance issues in nonlinear, time-varying system [3]. To overcome these challenges, advanced control techniques have been developed to address uncertainties and disturbances effectively.

Adaptive control approaches have been widely investigated to improve AUV performance in uncertain conditions. For instance [4] introduced terminal sliding mode control for underactuated AUVs, demonstrating accurate trajectory tracking. Similarly [5] presented a modified C/ GMRES- based predictive controller for efficient nonlinear tracking. In certain studies, the

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integration of adaptive control with intelligent control has been suggested, including approaches like Neural Network-based Control (NNC) [6], [7], Fuzzy Logic Controllers (FTC) [8] and Reinforcement Learning (RL) [9], [10].

Sliding Mode Control (SMC) is recognized for its robustness in handling external disturbances and system uncertainties. High- Order Sliding Mode Control (HOSMC) [18] techniques further address the chattering issue observed in conventional SMC methods, providing smoother control signals [11], [12]. For example [13] proposed a robust adaptive sliding mode controller to manage uncertain dynamics effectively. The controller is designed using dual closed-loop control approach, where the outer loop determines virtual velocity commands, and the inner loop generates the actual control inputs. However, the paper notes that the stability analysis for the outer- loop and inner- loop systems is performed independently, potentially neglecting the interactions between the loops. This could result in suboptimal performance in practical applications. It further states that all tracking errors in the closed-loop systems are uniformly ultimately bounded, as demonstrated through Lyapunov stability analysis. This indicates that while the errors may not converge to zero in finite time, they will remain within a certain boundary over time.

Backstepping-based sliding mode control has shown promise for stabilizing complex nonlinear systems, particularly under external disturbances [14]. Recent advancements in adaptive control have led to improved accuracy in trajectory tracking. For example, Peng et al. [15] proposed a dual closed-loop MPC strategy to handle uncertainties which is the combination of backstepping and optimal control, while [16] developed an adaptive backstepping sliding mode controller for lightweight AUVs with input saturation. However, both studies assume that external disturbance is bound and predictable. This assumption may limit the approach's effectiveness in handling unbounded external perturbations, potentially impacting real-world implementation.

Although these methods have demonstrated success, comprehensive comparisons of their performance under practical conditions remain limited. Notably, integrating super-twisting algorithms into backstepping controllers has shown significant improvements in robustness and chattering reduction [17]. However, the effectiveness of adaptive versus conventional backstepping super-twisting sliding mode controllers has yet to be fully explored.

This paper presents two advanced control strategies for trajectory tracking of the 6- DOF Xplorer- mini AUV, equipped with eight thrusters and developed at the Autonomous Marine Research Lab (AMARR Lab) at Faculty of Engineering, Kasetsart University, as shown in Fig. 1. The proposed strategies are the Backstepping Super-Twisting Sliding Mode Control (BSTSMC) and the Adaptive Backstepping Super- Twisting Sliding Mode Control (ABSTSMC). The ABSTSMC enhances the robustness of the BSTSMC by incorporating adaptive backstepping to address system nonlinearities and uncertainties, particularly under thruster faults. The key contributions of this study are:

1. A detailed comparative analysis of the two approaches in terms of tracking accuracy,

disturbance rejection, and robustness, especially under a thruster fault.

2. Stability analysis via Lyapunov function.

The structure of this paper is as follows: Section II presents the mathematical modeling of the AUV. Section III provides the design details of the proposed BSTSMC and ABSTSMC approach. Section IV discusses the simulation results, and Section V concludes the paper with final remarks and our future research work.

### II. MODELING

The mathematical model of an AUV is described in two coordinate frames: the body-fixed frame and earth-fixed frame. The kinematic and dynamic equations of an AUV in 6-DOF can be written as follows [1]:

$$\dot{\boldsymbol{\eta}} = \boldsymbol{J}(\boldsymbol{\eta})\boldsymbol{\nu}$$
$$\boldsymbol{M}_{RB}\dot{\boldsymbol{\nu}} + \boldsymbol{M}_{A}\dot{\boldsymbol{\nu}}_{r} + \boldsymbol{C}_{RB}(\boldsymbol{\nu})\boldsymbol{\nu} + \boldsymbol{C}_{A}(\boldsymbol{\nu}_{r})\boldsymbol{\nu}_{r} + \boldsymbol{D}(\boldsymbol{\nu}_{r})\boldsymbol{\nu}_{r} \quad (1)$$
$$+ \boldsymbol{g}(\boldsymbol{\eta}) = \boldsymbol{\tau} + \boldsymbol{\tau}_{dist}$$

where  $\mathbf{v} = [u, v, w, p, q, r]^T$  is the AUV linear and angular velocity vector relative to the body-fixed frame,  $\eta =$  $[x, y, z, \phi, \theta, \psi]^T$  is the AUV position and orientation vector relative to the earth-fixed frame,  $\boldsymbol{\nu}_r = \boldsymbol{\nu} - \boldsymbol{\nu}_c$  is the relative velocity of the AUV,  $v_c$  is the irrotational ocean current velocity relative to the body-fixed frame,  $M_{RB}$  is the rigid-body inertia matrix,  $M_A$  is the added mass matrix from hydrodynamic effects,  $C_{RB}(\nu)$  is the rigid-body Coriolis and centripetal matrix,  $C_A(\nu_r)$  is the hydrodynamic Coriolis and centripetal matrix,  $D(v_r)$  is the hydrodynamic damping matrix,  $g(\eta)$  is the restoring force and moment vector from the gravitational and buoyancy forces,  $\boldsymbol{\tau}$  is the control force and moment vector generated by thrusters,  $\tau_{dist}$  is the disturbance force and moment vector, and  $J(\eta)$  is the transformation matrix given by:

$$\boldsymbol{J}(\boldsymbol{\eta}) = \begin{bmatrix} \boldsymbol{R}_b^n(\boldsymbol{\Theta}_{nb}) & \boldsymbol{0} \\ \boldsymbol{0} & \boldsymbol{T}(\boldsymbol{\Theta}_{nb}) \end{bmatrix}$$
(2)

where  $\mathbf{\Theta}_{nb} = [\phi, \theta, \psi]^T$  is the AUV orientation described by the Euler angles,  $\mathbf{R}_b^n(\mathbf{\Theta}_{nb})$  is the rotation matrix, and  $T(\mathbf{\Theta}_{nb})$  is the transformation matrix.

The tracking problem is posted by giving  $\eta_d(t)$ , the AUV state vector is selected as  $[\eta, \dot{\eta}]^T$ . After the kinematic transformation process, the dynamic equation (1) can be written as:

$$M_{\eta}\ddot{\eta} + C_{\eta}\dot{\eta} + D_{\eta}\dot{\eta} + g_{\eta}$$
  
=  $\tau + \tau_{dist} + M_{A}\dot{\nu}_{c} + C_{A}(\nu_{r})\nu_{c}$  (3)  
+  $D(\nu_{r})\nu_{c}$ 

where

$$\begin{split} \boldsymbol{M}_{\eta} &= (\boldsymbol{M}_{RB} + \boldsymbol{M}_{A})\boldsymbol{J}^{-1}(\boldsymbol{\eta}) \\ \boldsymbol{C}_{\eta} &= [\boldsymbol{C}_{RB}(\boldsymbol{\nu}) + \boldsymbol{C}_{A}(\boldsymbol{\nu}_{r})]\boldsymbol{J}^{-1}(\boldsymbol{\eta}) - \boldsymbol{M}_{\eta}\boldsymbol{\dot{J}}(\boldsymbol{\eta})\boldsymbol{J}^{-1}(\boldsymbol{\eta}) \\ \boldsymbol{D}_{\eta} &= \boldsymbol{D}(\boldsymbol{\nu}_{r})\boldsymbol{J}^{-1}(\boldsymbol{\eta}) \\ \boldsymbol{g}_{\eta} &= \boldsymbol{g}(\boldsymbol{\eta}) \end{split}$$

## III. METHODOLOGY

In this section, we present two control methods combining backstepping with the Super-Twisting Algorithm (STA), a higher order sliding mode control introduced in [19]. The design of an adaptive control law based on the original Super-Twisting (STW) control is presented in [20] and further summarized in [21], [22]. Let  $u = \tau$  be the desired control force and moment vector from AUV controller. The proposed control u can be written as:

$$\boldsymbol{u} = \boldsymbol{u}_{nc} + \boldsymbol{u}_{sc} \tag{4}$$

where  $u_{nc}$  is the nominal control, and  $u_{sc}$  is the switching control that is used to handle disturbances and thruster faults.

# *A.* BSTSMC: Backstepping Super-Twisting Sliding Mode Control

Let  $x_1 = \eta$  and  $x_2 = \dot{\eta}$  is the AUV state vector. From (3), the AUV state-space model can be written as:

$$\dot{\mathbf{x}} = \begin{bmatrix} \dot{\mathbf{x}}_1 \\ \dot{\mathbf{x}}_2 \end{bmatrix} = \begin{bmatrix} \mathbf{x}_2 \\ \mathbf{a}(\mathbf{x}) + \mathbf{b}(\mathbf{x})\mathbf{u} + \mathbf{d} \end{bmatrix}$$
(5)

where

$$a(x) = (M_{\eta})^{-1} [-C_{\eta}\dot{\eta} - D_{\eta}\dot{\eta} - g_{\eta} + C_A(\nu_r)\nu_c + D(\nu_r)\nu_c]$$
  
$$b(x) = (M_{\eta})^{-1}$$
  
$$d = (M_{\eta})^{-1} [M_A\dot{\nu}_c - \tau_{dist}]$$

Define the position/orientation tracking error as:

$$\mathbf{z}_1 = \mathbf{x}_1 - \boldsymbol{\eta}_d \tag{6}$$

Then, using the backstepping method, the virtual velocity control  $\alpha_{virtual}$  is chosen as:

$$\boldsymbol{\alpha}_{virtual} = -\boldsymbol{K}_1 \boldsymbol{z}_1 + \boldsymbol{\dot{\eta}}_d \tag{7}$$

where  $K_1 \in \mathbb{R}^{6\times 6}$  is symmetric positive definite. Next, the virtual velocity tracking error  $\mathbf{z}_2$  is defined as:

$$\mathbf{z}_2 = \mathbf{x}_2 - \mathbf{\alpha}_{virtual} \tag{8}$$

From (5), (6), (7), and (8), the dynamic equation of  $\mathbf{z} = [\mathbf{z}_1, \mathbf{z}_2]^T$  can be written as:

$$\dot{\mathbf{z}} = \begin{bmatrix} \dot{\mathbf{z}}_1 \\ \dot{\mathbf{z}}_2 \end{bmatrix} = \begin{bmatrix} -K_1 \mathbf{z}_1 + \mathbf{z}_2 \\ \mathbf{a}(\mathbf{x}) + \mathbf{b}(\mathbf{x})\mathbf{u} + \mathbf{d} - \dot{\alpha}_{virtual} \end{bmatrix}$$
(9)

Consider the sliding variable:

$$\boldsymbol{\sigma} = \boldsymbol{K}_2 \boldsymbol{z}_1 + \boldsymbol{z}_2 \tag{10}$$

where  $K_2 \in \mathbb{R}^{6 \times 6}$  is a diagonal matrix, whose elements are positive. The time derivative of the sliding variable is given by:

$$\dot{\boldsymbol{\sigma}} = \boldsymbol{K}_2 \dot{\boldsymbol{z}}_1 + \dot{\boldsymbol{z}}_2 \tag{11}$$

The Lyapunov candidate function is chosen as:

$$V = \frac{1}{2} \boldsymbol{z}_1^T \boldsymbol{z}_1 + \frac{1}{2} \boldsymbol{\sigma}^T \boldsymbol{\sigma}$$
(12)

Then, the time derivative of is given by:

$$\dot{V} = \mathbf{z}_1^T \dot{\mathbf{z}}_1 + \boldsymbol{\sigma}^T \dot{\boldsymbol{\sigma}}$$
(13)

From (9) and (11), (13) can be rearranged as:

$$\dot{V} = \mathbf{z}_1^T (-\mathbf{K}_1 \mathbf{z}_1 + \mathbf{z}_2) + \boldsymbol{\sigma}^T (\mathbf{K}_2 \dot{\mathbf{z}}_1 + \boldsymbol{a}(\mathbf{x}) + \boldsymbol{b}(\mathbf{x}) \boldsymbol{u} \qquad (14) + \boldsymbol{d} - \dot{\boldsymbol{\alpha}}_{\text{virtual}})$$

From (4) and (14),  $\boldsymbol{u}_{nc}$  is given by:

$$\boldsymbol{u}_{nc} = \left(\boldsymbol{b}(\boldsymbol{x})\right)^{-1} \begin{bmatrix} -K_3 \boldsymbol{\sigma} - K_2^{-1} \boldsymbol{z}_2 - K_2 \dot{\boldsymbol{z}}_1 - \boldsymbol{a}(\boldsymbol{x}) \\ + \dot{\boldsymbol{\alpha}}_{virtual} \end{bmatrix}$$
(15)

where  $K_3 \in \mathbb{R}^{6 \times 6}$  is diagonal matrix with positive elements. The switching control  $u_{sc}$  is designed as:

$$\boldsymbol{u}_{sc} = (\boldsymbol{b}(\boldsymbol{x}))^{-1} \left[ -\boldsymbol{\alpha} |\boldsymbol{\sigma}|^{1/2} \operatorname{sign}(\boldsymbol{\sigma}) + \boldsymbol{\nu} \right]$$
  
$$\dot{\boldsymbol{\nu}} = -\boldsymbol{\beta} \operatorname{sign}(\boldsymbol{\sigma})$$
(16)

where  $|\boldsymbol{\sigma}|^{1/2} \operatorname{sign}(\boldsymbol{\sigma}) = [|\sigma_1|^{1/2} \operatorname{sign}(\sigma_1), ..., |\sigma_6|^{1/2} \operatorname{sign}(\sigma_6)]^T$ and  $\operatorname{sign}(\boldsymbol{\sigma}) = [sgn(\sigma_1), ..., sgn(\sigma_6)]^T$ .

From (15) and (16), (14) can be rearranged as:

$$\dot{V} = -\mathbf{z}_{1}^{T}\mathbf{K}_{1}\mathbf{z}_{1} - \mathbf{z}_{2}^{T}\mathbf{K}_{2}^{-1}\mathbf{z}_{2} - \boldsymbol{\sigma}^{T}\mathbf{K}_{3}\boldsymbol{\sigma} - \boldsymbol{\sigma}^{T} \begin{bmatrix} \boldsymbol{\alpha} | \boldsymbol{\sigma} |^{\frac{1}{2}} \operatorname{sign}(\boldsymbol{\sigma}) + \int_{0}^{t} \boldsymbol{\beta} \operatorname{sign}(\boldsymbol{\sigma}(\tau)) d\tau - d \end{bmatrix} \dot{V} = -\mathbf{z}_{1}^{T}\mathbf{K}_{1}\mathbf{z}_{1} - \mathbf{z}_{2}^{T}\mathbf{K}_{2}^{-1}\mathbf{z}_{2} - \boldsymbol{\sigma}^{T}\mathbf{K}_{3}\boldsymbol{\sigma} - \sum_{i=1}^{6} \begin{bmatrix} \alpha_{i} | \sigma_{i} |^{1/2} | \sigma_{i} | \\+ \beta_{i}\sigma_{i} \int_{0}^{t} \operatorname{sign}(\sigma_{i}(\tau)) d\tau - d_{i}\sigma_{i} \end{bmatrix}$$

where  $\boldsymbol{\alpha} = \operatorname{diag}(\alpha_1, \alpha_2, \dots, \alpha_6), \quad \alpha_i > 0 \quad \text{and} \quad \boldsymbol{\beta} = \operatorname{diag}(\beta_1, \beta_2, \dots, \beta_6), \quad \beta_i > 0.$ 

Under the assumption that  $|d_i| \le \delta_i |\sigma_i|^{1/2}$  for some unknown  $\delta_i > 0$ , and by selecting  $\alpha_i > \delta_i$ , we have:

$$d_i \sigma_i \le |d_i \sigma_i| \le \delta_i |\sigma_i|^{1/2} |\sigma_i| < \alpha_i |\sigma_i|^{1/2} |\sigma_i|$$

With appropriate tuning of  $\alpha_i$  and  $\beta_i$ , and under finite-time operation condition, we can conclude that:

$$\sum_{i=1}^{6} \left[ \alpha_i |\sigma_i|^{1/2} |\sigma_i| + \beta_i \sigma_i \int_0^t \operatorname{sign}(\sigma_i(\tau)) d\tau - d_i \sigma_i \right] > 0$$

Consequently,  $\dot{V}$  is negative- definite. By Barbalat's Lemma, the tracking error converges asymptotically to zero.

*B. ABSTSMC: Adaptive Backstepping Super-Twisting Sliding Mode Control* 

For the ABSTSMC, the adaptive law is given by:

$$\dot{\alpha}_{i} = \begin{cases} \omega_{1i} \sqrt{\frac{\gamma_{1i}}{2}}, & \text{if } \boldsymbol{\sigma} \neq 0\\ 0, & \text{if } \boldsymbol{\sigma} = 0 \end{cases}$$

$$\beta_{i} = 2\varepsilon_{i}\alpha_{i} + \lambda_{i} + 4\varepsilon_{i}^{2}$$

$$(17)$$

where  $\omega_{1i}$ ,  $\gamma_{1i}$ ,  $\varepsilon_i$  and  $\lambda_i$  are arbitrary positive constants. This ABSTSMC makes  $\boldsymbol{\sigma}$  and  $\boldsymbol{\dot{\sigma}}$  go to zero in finite time [19].

Now, the finite-time convergence of  $\boldsymbol{\sigma}$  and  $\dot{\boldsymbol{\sigma}}$  is proved. We rearrange (11) using (9), (15) and (16) as follows:

$$\dot{\boldsymbol{\sigma}} = (-K_3\boldsymbol{\sigma} - K_2^{-1}\boldsymbol{z}_2 + \boldsymbol{d}) + \boldsymbol{b}(\boldsymbol{x})\boldsymbol{u}_{sc}$$

which can be expressed as:

$$\dot{\boldsymbol{\sigma}} = \boldsymbol{\Psi}(\boldsymbol{z}, \boldsymbol{d}, t) + \boldsymbol{\Gamma}(\boldsymbol{z}, t) \boldsymbol{u}_{sw}$$
(18)

where  $\Psi(\mathbf{z}, \mathbf{d}, t) = -K_3 \sigma - K_2^{-1} \mathbf{z}_2 + \mathbf{d}$ ,  $\Gamma(\mathbf{z}, t) = \mathbf{I}_{6\times 6}$ and  $\mathbf{u}_{sw} = \mathbf{b}(\mathbf{x})\mathbf{u}_{sc}$ .

From (16), we further rearrange (18) into the form:

$$\dot{\boldsymbol{\sigma}} = -\boldsymbol{\alpha} |\boldsymbol{\sigma}|^{1/2} \operatorname{sign}(\boldsymbol{\sigma}) + \boldsymbol{\nu} + \boldsymbol{\Psi}(\boldsymbol{z}, \boldsymbol{d}, t)$$

$$\dot{\boldsymbol{\nu}} = -\boldsymbol{\beta} \operatorname{sign}(\boldsymbol{\sigma})$$
(19)

Then, (19) can be rewritten in scalar form as:

$$\dot{\sigma}_i = -\alpha_i |\sigma_i|^{1/2} \operatorname{sign}(\sigma_i) + v_i + \Psi_i(\mathbf{z}, \mathbf{d}, t)$$
  
$$\dot{v}_i = -\beta_i \operatorname{sign}(\sigma_i)$$
(20)

Consider the following Lyapunov candidate function [19]:

$$V_{2} = \sum_{i=1}^{6} V_{2i} = \sum_{i=1}^{6} \left[ V_{1i} + \frac{1}{2\gamma_{1i}} (\alpha_{i} - \alpha_{i}^{*})^{2} + \frac{1}{2\gamma_{2i}} (\beta_{i} - \beta_{i}^{*})^{2} \right]$$
(21)

where  $\alpha_i^*, \beta_i^*, \gamma_{1i}$  and  $\gamma_{2i}$  are some positive constants, and  $V_{1i}$  is given by:

$$V_{1i} = \boldsymbol{\zeta}_i^T \boldsymbol{P}_i \boldsymbol{\zeta}_i \tag{22}$$

where  $\boldsymbol{\zeta}_{i}^{T} = [|\sigma_{i}|^{1/2} \operatorname{sign}(\sigma_{i}), v_{i}]$  and  $\boldsymbol{P}_{i}$  is given by:

$$\boldsymbol{P}_{i} = \boldsymbol{P}_{i}^{T} = \begin{bmatrix} \lambda_{i} + 4\varepsilon_{i}^{2} & -2\varepsilon_{i} \\ -2\varepsilon_{i} & 1 \end{bmatrix} > 0$$
(23)

To ensure that derivative of  $V_{1i}$  is negative-definite,  $\alpha_i$  must satisfy the following inequality:

$$\alpha_i > \frac{\varepsilon_i \delta_i + (\lambda_i + 4\varepsilon_i^2)(2\varepsilon_i + \delta_i) + \varepsilon_i}{\lambda_i}$$
(24)

and  $\beta_i$  must satisfy (17). Then, the time derivative of  $V_{2i}$  can be expressed as (see (29) and (30) in [19]):

$$\dot{V}_{2i} \le -\kappa_i \sqrt{V_{2i}} + \xi_i \tag{25}$$

where  $\kappa_i = \min(\rho_i, \omega_{1i}, \omega_{2i})$  and  $\rho_i = \frac{2\varepsilon_i \lambda_{\min}^{1/2}(P_i)}{\lambda_{\max}(P_i)}$ , where  $\lambda_{\min}(P_i)$  and  $\lambda_{\max}(P_i)$  is minimum and maximum eigenvalue of matrix  $P_i$ , respectively. Here,  $\xi_i$  is given by:

$$\xi_{i} = -|\alpha_{i} - \alpha_{i}^{*}| \left(\frac{1}{\gamma_{1i}}\dot{\alpha}_{i} - \frac{\omega_{1i}}{\sqrt{2\gamma_{1i}}}\right) - |\beta_{i} - \beta_{i}^{*}| \left(\frac{1}{\gamma_{2i}}\dot{\beta}_{i} - \frac{\omega_{2i}}{\sqrt{2\gamma_{2i}}}\right)$$
(26)

We must assure that  $\xi_i = 0$ , which is to be achieved through adaptation of the gains  $\alpha_i$  and  $\beta_i$ , defined as follows:

$$\dot{\alpha}_{i} = \omega_{1i} \sqrt{\frac{\gamma_{1i}}{2}}$$

$$\dot{\beta}_{i} = \omega_{2i} \sqrt{\frac{\gamma_{2i}}{2}}$$
(27)

Therefore, for the finite time convergence of  $\boldsymbol{\sigma}$  and  $\dot{\boldsymbol{\sigma}}$ ,  $\alpha_i$  must satisfy inequality (24) via the adaptive equation in (27). By choosing

$$\varepsilon_i = \frac{\omega_{2i}\sqrt{\gamma_{2i}}}{2\omega_{1i}\sqrt{\gamma_{1i}}} \tag{28}$$

the ABSTSMC proposed in (17) is validated. Consequently, the tracking error converges asymptotically to zero.

#### IV. SIMULATION RESULTS

In this section, the AUV model and controller are simulated in ROS2 using Gazebo, as shown in Fig. 2, to evaluate performance under a thruster fault condition. The 3D spiral curve is derived from the equations presented in [6], and is expressed as follows:

$$\begin{aligned} x_{d}(t) &= x_{d}(0) + U_{d} \big( 1 - \cos(\psi_{d}(t)) \big) \\ y_{d}(t) &= y_{d}(0) + U_{d} \sin(\psi_{d}(t)) \\ z_{d}(t) &= z_{d}(0) + w_{d} t \end{aligned}$$

$$\begin{split} \phi_d(t) &= 0\\ \theta_d(t) &= 0\\ \psi_d(t) &= \psi_d(0) + r_d t \end{split}$$

where  $U_d = 4$ ,  $w_d = -0.06$ ,  $r_d = 0.2$ ,  $x_d(0) = 0$ ,  $z_d(0) = -0.2$ , and  $\psi_d(0) = 0$ .

The desired trajectory  $\eta_d$  and the actual trajectories  $\eta_s$  under the BSTSMC and ABSTSMC are shown in Fig. 3, 4, and 5. The thrust forces for each method are shown in Fig. 6 and 7.



Fig.2 AUV Gazebo simulation

The tracking performance of the BSTSMC and ABSTSMC is evaluated using two metrics: Maximum Absolute Error (MaxAE) and Mean Integral Absolute Error (MIAE):

$$MaxAE = \max_{\substack{t \in [T_1, T_2] \\ t \in [T_2, T_2]}} |\eta_i(t) - \eta_{d,i}(t)|$$
(29)

$$\text{MAE} = \frac{1}{T_2 - T_1} \int_{T_1}^{T_2} \left| \eta_i(t) - \eta_{d,i}(t) \right| dt \qquad (30)$$

where  $[T_1, T_2]$  represents the interval for performance evaluation,  $\eta_i(t)$  denotes the actual position and orientation of the AUV,  $\eta_{d,i}(t)$  is the desired position and orientation of the AUV, with i = 1, 2, 3, 4, 5, 6.

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Fig. 3 AUV trajectories for the 3D spiral curve

The tracking error performance of the BSTSMC and ABSTSMC is evaluated over the time interval of 0 to 165 seconds using the same metrics, as shown in Tables I and II. The steady-state tracking error, evaluated during the

period from 165 to 180 seconds after the AUV has settled, is presented in Tables III and IV.

The maximum absolute values and mean integral absolute values of thrust forces generated from eight thrusters are shown in Table V and VI. The parameters in STA with adaptive gains were set to  $\boldsymbol{\epsilon} = 0.01 \times I_{6\times 6}$ ,  $\boldsymbol{\lambda} = 0.1 \times I_{6\times 6}$ ,  $\boldsymbol{\gamma}_1 = 0.1 \times I_{6\times 6}$ , and  $\boldsymbol{\omega}_1 = 0.01 \times I_{6\times 6}$  where  $I_{6\times 6}$  is the  $6\times 6$  identity matrix.



Fig. 4 AUV position for the 3D spiral curve: (a) Position along the xaxis, (b) Position along the y-axis, (c) Position along the z-axis.





(c)

75 100 time (s) 125

150

175

Fig. 5 AUV orientation for the 3D spiral curve: (a) Orientation about the x-axis, (b) Orientation about the y-axis, (c) Orientation about the z-axis.



Fig. 6 Thrust forces generated by thrusters #1 to #4 for the 3D spiral curve: (a) Thruster #1, (b) Thruster #2, (c) Thruster #3, (d) Thruster #4.



Fig. 7 Thrust forces generated by thrusters #5 to #8 for the 3D spiral Curve: (a) Thruster #5, (b) Thruster #6, (c) Thruster #7, (d) Thruster #8.

TRACKING ERRORS (MAXAE)			
position and orientation	BSTSMC	ABSTSMC	% relative to BSTSMC
<i>x</i> ( <i>m</i> )	0.0939	0.0593	-36.85
y (m)	0.1037	0.1170	12.83
z (m)	0.1465	0.1161	-20.75
φ (degree)	10.3382	9.1756	-11.25
θ (degree)	4.5049	3.5918	-20.27
$\psi$ (degree)	19.8106	21.5047	8.55

TABLEI

TABLE II TRACKING ERRORS (MIAE)

IRACKING ERRORS (MIAE)			
position and orientation	BSTSMC	ABSTSMC	% relative to BSTSMC
<i>x</i> ( <i>m</i> )	0.0299	0.0119	-60.20
y (m)	0.0352	0.0146	-58.52
z (m)	0.0808	0.0564	-30.20
φ (degree)	5.2471	3.7845	-27.87
$\theta$ (degree)	0.9716	1.4350	47.69
ψ (degree)	13.3509	13.6151	1.98

TABLE III

STEADY-STATE TRACKING ERRORS (MAXAE)			
position and orientation	BSTSMC	ABSTSMC	% relative to BSTSMC
<i>x</i> ( <i>m</i> )	0.0113	0.0118	4.42
y (m)	0.0249	0.0053	-78.71
z (m)	0.0270	0.0279	3.33
φ (degree)	1.3454	1.9676	46.25
$\theta$ (degree)	0.6180	0.8425	36.33
$\psi$ (degree)	3.2292	4.5170	39.88

TABLE IV

STEADY-STATE TRACKING ERRORS (IMIAE)			
position and orientation	BSTSMC	ABSTSMC	% relative to BSTSMC
<i>x</i> ( <i>m</i> )	0.0082	0.0070	-14.63
y (m)	0.0211	0.0020	-90.52
z (m)	0.0247	0.0247	0.00
$\phi$ (degree)	0.9641	1.3361	38.59
θ (degree)	0.4564	0.6207	36.00
ψ (degree)	1.5016	1.7521	16.68

TABLE V

MAX ABSOLUTE VALUE OF THRUSTER FORCES (N)			
Thruster No.	BSTSMC	ABSTSMC	% relative to BSTSMC
1	19.2739	24.5171	27.2
2	14.9668	18.7160	25.05
3 (Fault)	-	-	-
4	16.4010	14.3324	-12.61
5	11.8590	12.9679	9.35
6	11.4670	13.2923	15.92
7	12.7368	13.8536	8.77
8	18.7710	19.1895	2.23

TABLE VI	
MEAN INTEGRAL ABSOLUTE VALUE OF THRUSTER FORCES (N)	

Thruster No.	BSTSMC	ABSTSMC	% relative to BSTSMC
1	3.0246	3.2851	8.61
2	3.3834	3.5443	4.76
3 (Fault)	-	-	-
4	5.6551	5.1746	-8.5
5	4.8100	5.1457	6.98
6	5.5218	5.8199	5.4
7	1.7725	2.0066	13.21
8	1.8047	2.0819	15.36

Tables I and II show that ABSTSMC achieves better performance than BSTSMC in position tracking under unknown perturbations, based on the MaxAE and MIAE of tracking errors. Conversely, as indicated in Tables III and IV, BSTSMC performs more effectively in orientation tracking when considering the MaxAE and MIAE of steady-state errors. Additionally, Tables V and VI reveal that in the event of a failure in thruster #3, the overall forces of all thrusters, except for thruster #4, increase in the ABSTSMC due to gain adjustments.

## V. CONCLUSION

This research presents comparative analysis of BSTSMC and ABSTSMC approaches for controlling AUV under thruster fault conditions in the presence of ocean currents. Simulation results show that both BSTSMC and ABSTSMC enable AUV to follow a reference trajectory accurately, even when faced with unexpected disturbances and thruster faults.

In future work, we plan to conduct pool tests of the Xplorer-mini AUV using both control algorithms to evaluate their performance under real-world conditions.

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